

405(5): Solution of Differential Equation for $\langle \underline{s}_r \cdot \underline{s}_r \rangle$

This is eq. (1) of Note 405(4):

$$\Delta \phi = \frac{r^2}{2} \left(\frac{\omega}{r} - \frac{d\omega}{dr} \right) \quad - (1)$$

$$= \frac{4}{3} \frac{\langle \underline{s}_r \cdot \underline{s}_r \rangle}{r^2} - \frac{1}{3r} \frac{d}{dr} \langle \underline{s}_r \cdot \underline{s}_r \rangle$$

If $f(r) := \langle \underline{s}_r \cdot \underline{s}_r \rangle - (2)$

Eq. (1) can be written as:

$$4f(r) - r \frac{df(r)}{dr} = 3r^2 \Delta \phi. \quad - (3)$$

According to the Wolfram site a differential equation
the solution of Eq. (3) is:

$$\langle \underline{s}_r \cdot \underline{s}_r \rangle = \frac{3}{2} r^2 \Delta \phi + C_1 r^4 \quad - (4)$$

where C_1 is a constant of integration. If C_1
is chosen to be zero then:

$$\boxed{\frac{\langle \underline{s}_r \cdot \underline{s}_r \rangle}{r^2} = \frac{3}{2} \Delta \phi} \quad - (5)$$

which shows that any precession is due to the
isotropically averaged vacuum fluctuation $\langle \underline{s}_r \cdot \underline{s}_r \rangle$.

The starting equation is:

$$\underline{F} = -m \underline{\nabla} \phi + m \underline{\omega} \phi \quad - (6)$$

and this is solved in the quasi-al approximation

2) for a nearly circular orbit.

If each orbit is considered to be generated by a Thomas precession of velocity:

$$v = \omega_0 r \quad - (7)$$

then

$$\Delta \phi \sim \pi \left(\frac{v}{c} \right)^2 \quad - (8)$$

From eqs. (5) and (8):

$$\boxed{\frac{\langle \underline{\delta r} \cdot \underline{\delta r} \rangle}{r^2} = \frac{3}{2} \pi \left(\frac{v}{c} \right)^2} \quad - (9)$$

Application to Gravity Probe B.

From eq. (21) of the previous note the velocities of the gravitational, de Sitter and Lense Thirring precessions are:

$$\left. \begin{aligned} v(\text{grav}) &= 1.84 \times 10^5 \text{ m s}^{-1} \\ v(\text{de Sitter}) &= 1.25 \times 10^5 \text{ m s}^{-1} \\ v(\text{Lense Thirring}) &= 5.70 \times 10^3 \text{ m s}^{-1} \end{aligned} \right\} - (10)$$

So:

$$\left(\frac{\langle \underline{\delta r} \cdot \underline{\delta r} \rangle}{r^2} \right)_{\text{grav.}} = 1.78 \times 10^{-6} \quad - (11)$$

$$\left(\frac{\langle \underline{\delta r} \cdot \underline{\delta r} \rangle}{r^2} \right)_{\text{de Sitter}} = 8.19 \times 10^{-7} \quad - (12)$$

$$\left(\frac{\langle \underline{\delta r} \cdot \underline{\delta r} \rangle}{r^2} \right)_{\text{Lense Thirring}} = 1.70 \times 10^{-9} \quad - (13)$$