

377(1): Orbital Equations with Relativistic Form of Newton's Law

In this case the force is defined by

$$\underline{F} = \frac{d}{dt} (\gamma m \underline{v}_0) = \underline{dp} \quad - (1)$$

The relativistic kinetic energy is derived from this force:

$$W = T = \int \frac{d}{dt} (\gamma m \underline{v}_0) \cdot \underline{v}_0 dt \quad - (2)$$

where W is the work done. So:

$$T = m \int_0^{v_0} v_0 d(\gamma v_0) \quad - (3)$$

and integrating by parts:

$$T = \gamma m v_0^2 - m \int_0^{v_0} \frac{v_0 dv_0}{(1 - v_0^2/c^2)^{1/2}} \quad - (4)$$

$$= \gamma m v_0^2 + m c^2 (1 - v_0^2/c^2) \Big|_0^{v_0}$$

$$= \gamma m v_0^2 + m c^2 (1 - v_0^2/c^2)^{1/2} - m c^2$$

$$= (\gamma - 1) m c^2$$

This procedure gives the total energy:

$$E = T + m c^2 = \gamma m c^2 \quad - (5)$$

and the Hamiltonian:

$$H = \gamma m c^2 + \bar{U} \quad - (6)$$

and the Lagrangian:

$$L = -mc^2 \left(1 - \frac{v_0^2}{c^2}\right)^{-1/2} - U \quad (7)$$

$$= -\frac{mc^2}{\gamma} - U \quad (8)$$

So the force that corresponds to eqs (6) and (8) is:

$$F = \frac{d}{dt} (\gamma m v_0) = m \frac{d}{dt} (\gamma v_0)$$

$$= m \left(v_0 \frac{d\gamma}{dt} + \gamma \frac{dv_0}{dt} \right)$$

$$= m \left(v_0 \frac{d\gamma}{dv_0} \frac{dv_0}{dt} + \gamma \frac{dv_0}{dt} \right)$$

$$= m \frac{dv_0}{dt} \left(v_0 \frac{d\gamma}{dv_0} + \gamma \right) \quad (9)$$

Now use: $\gamma = \left(1 - \frac{v_0^2}{c^2}\right)^{-1/2} \quad (10)$

so $\frac{d\gamma}{dv_0} = \frac{v_0}{c^2} \left(1 - \frac{v_0^2}{c^2}\right)^{-3/2} = \gamma^3 \frac{v_0}{c^2} \quad (11)$

So $F = m \gamma \frac{dv_0}{dt} \left(1 + \gamma^2 \frac{v_0^2}{c^2}\right) \quad (12)$

$$= m \gamma \frac{dv_0}{dt} \left(1 + \left(1 - \frac{v_0^2}{c^2}\right)^{-1} \frac{v_0^2}{c^2}\right)$$

$$= m \gamma \frac{dv_0}{dt} \left(\frac{1 + \frac{v_0^2}{c^2}}{c^2 \left(1 - \frac{v_0^2}{c^2}\right)} \right)$$

$$= m \gamma \frac{dv_0}{dt} \left(\frac{c^2 \left(1 - \frac{v_0^2}{c^2}\right) + v_0^2}{c^2 \left(1 - \frac{v_0^2}{c^2}\right)} \right)$$

$$= m \gamma^3 \frac{dv_0}{dt}$$

Therefore the orbital equation is :

$$\gamma^3 \ddot{\underline{r}} = -mG \frac{\underline{r}}{r^3} \quad \text{--- (13)}$$

$$\text{--- (14)}$$

i.e.

$$\ddot{x} = -mG \left(1 - \frac{\dot{x}^2 + \dot{y}^2}{c^2}\right)^{3/2} \frac{x}{(x^2 + y^2)^{3/2}}$$

$$\ddot{y} = -mG \left(1 - \frac{\dot{x}^2 + \dot{y}^2}{c^2}\right)^{3/2} \frac{y}{(x^2 + y^2)^{3/2}} \quad \text{--- (15)}$$

The components of acceleration are:

$$g_x = \frac{\ddot{x}}{\left(1 + \frac{\dot{x}^2 + \dot{y}^2}{c^2}\right)^{3/2}} = -mG \frac{x}{(x^2 + y^2)^{3/2}} \quad \text{--- (16)}$$

$$\text{and } g_y = \frac{y}{\left(1 + \frac{\dot{x}^2 + \dot{y}^2}{c^2}\right)^{3/2}} = -MG \frac{y}{(x^2 + y^2)^{3/2}} \quad - (17)$$

The relevant field equations are:

$$\underline{\nabla} \times \underline{g} = \underline{0} \quad - (18)$$

$$\underline{\nabla} \cdot \underline{g} = \underline{\kappa} \cdot \underline{g} = 4\pi G \rho \quad - (19)$$

$$\frac{d\underline{g}}{dt} = \underline{0} \quad - (20)$$

Eq. (19) gives:

$$\rho_M = \frac{M}{2\pi(x^2 + y^2)^{3/2}} \quad - (21)$$

$$\text{and } \frac{dg_x}{dx} + \frac{dg_y}{dy} = \frac{2MG}{(x^2 + y^2)^{3/2}} \quad - (22)$$

$$= \kappa_x g_x + \kappa_y g_y$$

$$= \kappa_x \ddot{X} + \kappa_y \ddot{Y}$$

$$\frac{\left(1 + \frac{\dot{x}^2 + \dot{y}^2}{c^2}\right)^{3/2}}{c^2}$$

$$= -MG \left(\frac{\kappa_x X + \kappa_y Y}{(x^2 + y^2)^{3/2}} \right)$$

So derive two more equations:

$$X\kappa_x + Y\kappa_y = -2 \quad - (23)$$

5) and

$$k_x g_x + k_y g_y = 4\pi G \frac{\rho}{M} = \frac{k_x \ddot{X} + k_y \ddot{Y}}{\left(1 + \frac{\dot{X}^2 + \dot{Y}^2}{c^2}\right)^{3/2}} \quad (24)$$

Eqs. (14), (15), (23) and (24) must be solved simultaneously with input parameters k_x and k_y .

Note that eqs (14) and (15) are possible solutions of eq. (24), the solutions:

$$k_x g_x = \frac{k_x \ddot{X}}{\left(1 + \frac{\dot{X}^2 + \dot{Y}^2}{c^2}\right)^{3/2}} \quad (25)$$

and

$$k_y g_y = \frac{k_y \ddot{Y}}{\left(1 + \frac{\dot{X}^2 + \dot{Y}^2}{c^2}\right)^{3/2}} \quad (26)$$

However, a more general solution is found from eq. (24) by using k_x and k_y as input parameter. Eq (24) is then solved simultaneously with eqs. (14) and (15) to give all kinds of precessing orbits.

The Minkowski four force is:

$$F^\mu = dp^\mu / d\tau \quad (27)$$

where

$$p^\mu = \left(\frac{E}{c}, \underline{p}\right) \quad (28)$$

is the relativistic four momentum. Here:

$$E = \gamma mc^2, \quad \underline{p} = \gamma m \underline{v}_0 \quad - (29)$$

The vector part of the Minkowski four-force is:

$$\underline{F}(\text{Minkowski}) = \gamma \underline{F} \quad - (30)$$

where \underline{F} is defined in eq. (1).

It is clear that eq. (1) is the force definition that leads to the Lagrangian (8) and Hamiltonian (7). So far self consistency of orbital force equation is (13). The Minkowski four force is defined with the proper time, which the particle's own time, i.e. the time in the frame attached to the particle. Therefore eq. (13) in the particle frame is:

$$\gamma^4 \ddot{\underline{r}} = -M G \frac{\underline{r}}{r^3} \quad - (31)$$

In the stationary or observer frame, r^3 the equations to be solved are:

$$\gamma^3 \ddot{\underline{r}} = -M G \frac{\underline{r}}{r^3} \quad - (32)$$

together with the field equations in the stationary or observer frame, eqs. (18) to (20).